(2) All other switches (except ignition switch) -- OFF

(3) Check condition of circuit breakers to identify faulty circuit if possible. Leave faulty circuit deactivated.

(4) Master Switch -- ON.

(5) Select switches ON successively, permitting a short time delay to elapse after each switch is turned on until the short circuit is localized.

(6) Make sure fire is completely extinguished before opening ventilators.

FLIGHT IN ICING CONDITIONS.

Although flying in known icing conditions is prohibited, an unexpected icing encounter should be handled as follows:

Turn on pitot heat switch (if installed).

(2) Turn back or change altitude to obtain an outside air temperature that is less conducive to icing.

(3) Pull cabin heat control full out to obtain windshield defroster airflow. Adjust cabin air control to get maximum defroster heat and airflow.

(4) Open the throttle to increase engine speed and minimize ice build up on propeller blades.

(5) Watch for signs of carburetor air filter ice and apply carburetor heat as required. An unexplained loss in engine speed could be caused by carburetor ice or air intake filter ice.

(6) Plan a landing at the nearest airport. With an extremely rapid ice build-up, select a suitable "off airport" landing site.

(7) With an ice accumulation of one quarter inch or more on the wing leading edges, be prepared for significantly higher stall speed.

(8) Leave wing flaps retracted. With a severe ice build-up on the horizontal tail, the change in wing wake airflow direction caused by wing flap extension could result in a loss of elevator effectiveness.

(9) Open left window and, if practical, scrape ice from a portion of the windshield for visibility in the landing approach.

(10) Perform a landing approach using a forward slip, if necessary, for improved visibility.

(11) Approach at 75 to 85 MPH, depending upon the amount of ice accumulation.

(12) Avoid steep turns during the landing approach.

(13) Perform a landing in level attitude.

Section IV

OPERATING LIMITATIONS

OPERATIONS AUTHORIZED.

Your Cessna exceeds the requirements of airworthiness as set forth by the United States Government, and is certificated under FAA Type Certificate No. 3A12 as Cessna Model No. 172M.

The airplane may be equipped for day, night, VFR, or IFR operation. Your Cessna Dealer will be happy to assist you in selecting equipment best suited to your needs.

Your airplane must be operated in accordance with all FAA-approved markings and placards in the airplane. If there is any information in this section which contradicts the FAA-approved markings and placards, it is to be disregarded.

MANEUVERS - NORMAL CATEGORY.

This airplane is certificated in both the normal and utility category. The normal category is applicable to airplanes intended for non-aerobatic operations. These include any maneuvers incidental to normal flying, stalls (except whip stalls) and turns in which the angle of bank is not more than 60°. In connection with the foregoing, the following gross weight and flight load factors apply:

Gross Weight								2300 lbs
Flight Load Factor *Flaps Up							+3.8	-1.52
*Flaps Down .								

*The design load factors are 150% of the above, and in all cases, the structure meets or exceeds design loads.

MANEUVERS - UTILITY CATEGORY.

This airplane is not designed for purely aerobatic flight. However, in the acquisition of various certificates such as commercial pilot, instrument pilot and flight instructor, certain maneuvers are required by the FAA. All of these maneuvers are permitted in this airplane when operated in the utility category. In connection with the utility category, the following gross weight and flight load factors apply, with maximum entry speeds for maneuvers as shown:

Gross Weight	- +							2000 lbs
Flight Load Factor								
Flaps Up							.+4.4	-1.76
Flaps Down .							.+3.0	

In the utility category, the baggage compartment and rear seat must not be occupied. No aerobatic maneuvers are approved except those list " ed below:

MANEUVER												NDED ENTRY SPEED
Chandelles.										٠		120 mph (104 knots)
												120 mph (104 knots)
Steep Turns												112 mph (97 knots)
Spins												. Slow Deceleration
Stalls (Excep	ot	W	nip	S	tal	ls)						. Slow Deceleration

^{*}Abrupt use of the controls is prohibited above 112 MPH.

Aerobatics that may impose high loads should not be attempted. The important thing to bear in mind in flight maneuvers is that the aircraft is clean in aerodynamic design and will build up speed quickly with the nose down. Proper speed control is an essential requirement for execution of any maneuver, and care should always be exercised to avoid excessive speed which in turn can impose excessive loads. In the execution of all maneuvers, avoid abrupt use of controls. Intentional spins with flaps extended are prohibited.

IRSPEED LIMITATIONS (CAS).

The following is a list of the certificated calibrated airspeed (CAS) limitations for the aircraft.

Never Exceed Speed (glide or dive, smooth air)			182 MPH
Maximum Structural Cruising Speed			145 MPH
Maximum Speed, Flaps Extended			100 MPH
*Maneuvering Speed			112 MPH

^{*}The maximum speed at which you may use abrupt control travel.

AIRSPEED INDICATOR MARKINGS.

The following is a list of the certificated calibrated airspeed markings (CAS) for the aircraft.

Never Exceed (glide or div	re,	SI	mo	oth	ai	r)	182 MPH (red line)
Caution Range							145-182 MPH (yellow arc)
Normal Operating Range				*			. 61-145 MPH (green arc)
Flap Operating Range							. 54-100 MPH (white arc)

ENGINE OPERATION LIMITATIONS.

Power and Speed 150 BHP at 2700 RPM

ENGINE INSTRUMENT MARKINGS.

BIL	PRESSURE GAGE.					
	Minimum Idling					25 psi (red line)
	Normal Operating Range					
	Maximum					100 ngi (rod line)

FUEL QUANTITY INDICATORS.

Empty (2.0 gallons unusable each tank) E (red line)

TACHOMETER.

CARBURETOR AIR TEMPERATURE GAGE (OPT).

Icing Range -15° to 5°C (yellow arc)

WEIGHT AND BALANCE.

The following information will enable you to operate your Cessna within the prescribed weight and center of gravity limitations. To figure the weight and balance for your particular airplane, use the Sample Problem, Loading Graph, and Center of Gravity Moment Envelope as follows:

Take the licensed empty weight and moment from the Weight and Balance and Installed Equipment Data sheet (or changes noted on FAA Form 337) carried in your airplane, and write them down in the column titled YOUR AIRPLANE on the Sample Loading Problem.

NOTE

The Weight and Balance and Installed Equipment Data sheet is included in the aircraft file. In addition to the licensed empty weight and moment noted on this sheet, the c.g. arm (fuselage station) is also shown, but need not be used on the Sample Loading Problem. The moment shown on the sheet must be divided by 1000 and this value used as the moment/1000 on the loading problem.

Use the Loading Graph to determine the moment/1000 for each additional item to be carried, then list these on the loading problem.

NOTE

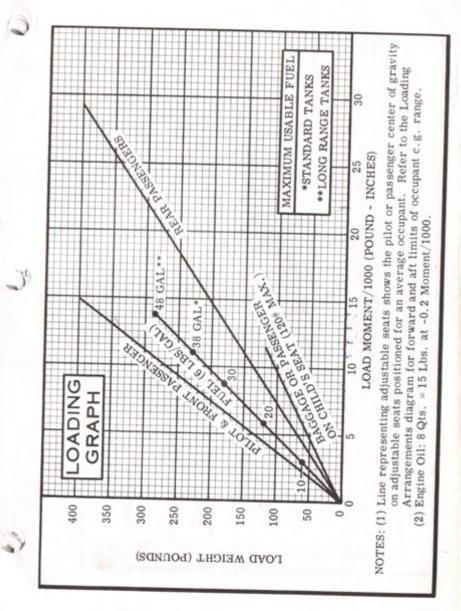
Loading Graph information is based on seats positioned for average occupants and baggage loaded in the center of the baggage area. For other than average loading situations, the Sample Loading Problem lists fuselage stations for these items to indicate their forward and aft c.g. range limitation (seat travel or baggage area limitation). Additional moment calculations, based on the actual weight and c.g. arm (fuselage station) of the item being loaded, must be made if the position of the load is different from that shown on the Loading Graph.

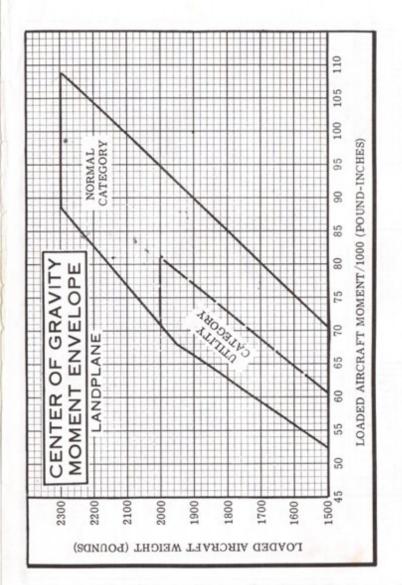
Total the weights and moments/1000 and plot these values on the Center of Gravity Moment Envelope to determine whether the point falls within the envelope, and if the loading is acceptable.

STATION STATION IC.G. ARMI (C.G. ARM) LOADING ARRANGEMENTS PILOT PASS PASS. (34 10 46) (34 18 41) * Putot or passenger center of gravity on adjustable seats positioned for average occupant. Numbers in paren-theses indicate forward and all limits REAR PASS. REAR PASS. of occupant center of gravity range ** Arm measured to the center of the CHILD SEAT BAGG. NOTE: The aft buggage wall (approximate station 108) can be used as a convenient interior reference point for determining the location of buggage area functage STANDARD OPTIONAL SEATING SEATING

		SAN	SAMPLE	Y	YOUR
(אוא	LAINE	בצוב	AIRPLAINE
ω	SAMPLE LOADING PROBLEM	Weight (lbs.)	Moment (lbins. /1000)	Weight (lbs.)	Moment (lbins. /1000
1.	Licensed Empty Weight (Sample Airplane)	1364	51.7	The con-	ż
63	Oil (8 qts The weight of full oil may be used for all calculations)	15	-0.2	15	-0.2
3	Fuel (Standard - 38 Gal at 6#/Gal)	228	10.9		
	Fuel (Long Range - 48 Gal at 6#/Gal)				
4	Pilot and Front Passenger (Station 34 to 46)	340	12.6		
10	Rear Passengers	340	24.8		-
.9	Baggage (or Passenger on Child's Seat) (Station 82 to 108)	13	1.2		
7.	TOTAL WEIGHT AND MOMENT	2300	101.0		

Locate this point (2300 at 101.0) on the center of gravity moment envelope, and since this point falls within the envelope, the loading is acceptable.





Section V

CARE OF THE AIRPLANE

If your airplane is to retain that new plane performance and dependability, certain inspection and maintenance requirements must be followed. It is wise to follow a planned schedule of lubrication and preventive maintenance based on climatic and flying conditions encountered in your locality.

Keep in touch with your Cessna Dealer and take advantage of his knowledge and experience. He knows your airplane and how to maintain it. He will remind you when lubrications and oil changes are necessary, and about other seasonal and periodic services.

GROUND HANDLING.

The airplane is most easily and safely maneuvered by hand with the tow-bar attached to the nose wheel. When towing with a vehicle, do not exceed the nose gear turning angle of 30° either side of center, or damage to the gear will result. If the airplane is towed or pushed over a rough surface during hangaring, watch that the normal cushioning action of the nose strut does not cause excessive vertical movement of the tail and the resulting contact with low hangar doors or structure. A flat nose wheel tire or deflated strut will also increase tail height.

MOORING YOUR AIRPLANE.

Proper tie-down procedure is your best precaution against damage to your parked airplane by gusty or strong winds. To tie down your airplane securely, proceed as follows:

(1) Set the parking brake and install the control wheel lock.

(2) Tie sufficiently strong ropes or chains (700 pounds tensile strength) to wing, tail and nose tie-down rings and secure each rope to a ramp tie-down.